



Article Wind Flow and Its Interaction with a Mobile Solar PV System Mounted on a Trailer

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Abstract: Efficient implementation of clean energy technologies is paramount, with mobile solar PV systems on trailers (MSPTs) emerging as pivotal solutions, particularly in regions with limited power grid access. This endeavour is vital for meeting escalating electricity demands and aligning with the UN Sustainable Development Goal (SDG), aimed at ensuring dependable and sustainable energy provision in developing countries. This study investigates the aerodynamic behaviour of a designed MSPT using numerical simulation and experimental methods, thereby offering optimization potential for MSPT design and enhancing overall performance and reliability. Specifically, the study focuses on the effects of wind velocity and tilt angles on the drag and lift forces, as well as drag and lift coefficients on the panel used in the MSPT system. The overall wind force on the entire MSPT, including nine large solar PV panels, is scrutinised, considering combined wind flow and system geometry effects. The numerical investigations were conducted using ANSYS-Fluent software (version 2022/R2) and experimental testing was performed within the C15-10 Wind Tunnel, utilizing scaled-down models to validate the accuracy of the simulation. The findings from the numerical investigations showed an increased turbulence caused by gaps between panels, resulting in almost 62% higher suction flow velocity and 22% higher suction pressure compared to a single panel. Drag and lift forces on the entire MSPT were approximately 6.7 and 7.8 times greater than those on a single panel with the same 30-degree tilt angle, respectively. The findings revealed that scaling forces on a single panel is insufficient for accurately predicting the aerodynamic forces on the entire MSPT. The insights and the knowledge from this study pave the way for further improvements in mobile solar PV technology.

Keywords: mobile solar PV system; aerodynamic; Computational Fluid Dynamics (CFD); wind effects; Sustainable Development Goal (SDG)

1. Introduction

The need for sustainable development, reduction in energy-related greenhouse gas emissions, and the desire to attain carbon-neutral economy have been major contributing factors in the drive to shift from fossil-based systems of energy production and consumption towards renewable and clean energy sources. Photovoltaic (PV) harvesting energy systems constitute one of the major technologies which could contribute to meeting the UN Sustainable Development Goal (SDG) on ensuring reliable and sustainable energy supply through effective sun energy harvesting to generate electricity while minimizing adverse



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environmental impacts [1]. This encompasses the introduction of the solar PV system as a key renewable energy source, feasible for electricity generation at convincing costs per kilowatt in diverse global regions [2,3].

Mobile solar PV systems play a crucial role in regions where access to the power grid is limited or unreliable [4]. These systems provide a sustainable and independent source of energy, particularly in remote areas or during emergencies such as in healthcare buildings. The mobility of solar PV systems allows them to be easily deployed and relocated as needed, making them suitable for various applications such as disaster relief efforts, outdoor events, and mobile installations, including recreational vehicles and boats. Advancements in battery storage technology have improved their effectiveness by enabling energy storage for use during non-sunlight hours. This ensures a consistent power supply even in offgrid locations. Despite the several advantages of the mobile solar PV systems, one of the challenging problems that needs to be addressed is ensuring the structural integrity of solar panels to withstand wind loads in various circumstances. Solving or reducing this challenge will have significant impacts on the deployment of the system.

2. Literature Review

The aerodynamics of solar panels are categorised into roof-mounted panels and ground-mounted, depending on the wind conditions and turbulence intensity [5]. The most popular method of installation is roof-mounted solar systems for generating clean and renewable energy. A solar PV system erected on a roof can be installed on flat or sloping roofs, including metal composite roofs or rubber materials. When installed on residential, commercial, or industrial buildings, they can provide significant cost savings on electricity bills and help to reduce greenhouse gas emissions [6,7]. Roof-mounted panels are influenced by surrounding walls and building edges, causing airflow separation and vortex generation, which may lead to a high-pressure zone on the roof [8]. The loads on the panels are determined by their inclination angle, affected by different factors including wind direction, panel position, and building height [9,10]. A ground-mounted solar system, on the other hand, is a free-standing solar system that is mounted on the ground. These solar systems are installed with a metal frame or a single pole. The same type of solar panels is used in ground-mounted solar PV systems as in rooftop systems, but they are installed differently. Ground-mounted solar panels, located within the boundary layer, require precise array configuration, wind direction, and wind speed and turbulence definitions for accurate load calculations [11,12].

The analytical investigation aims to utilise aerodynamic formulations to determine the forces experienced by the panels. The lift force represents the component of the aerodynamic force that acts perpendicular to the airflow and is mathematically described by Equation (1). Likewise, the drag force corresponds to the component of the aerodynamic force that acts tangentially to the airflow and can be expressed by Equation (2).

$$F_L = \frac{1}{2} A \rho C_L V^2 \tag{1}$$

$$F_D = \frac{1}{2} A \rho C_D V^2 \tag{2}$$

where F_L and F_D are lift and drag forces (N), respectively. *A* is the area (m²) of the panel subjected to the air flow, ρ is the mass density (kg/m³) of the air flow, and *V* is the velocity (m/s) of wind relative to the panel. C_L and C_D are the lift and drag coefficients, respectively.

Obtaining accurate values for C_L and C_D is crucial when calculating the aerodynamic forces on the panels analytically. These coefficients need to be determined specifically for each panel size and tilting position. The most reliable method for determining these coefficients is using experimental tests. Experimental tests allow the determination of aerodynamic forces, drag coefficient, lift coefficient, and other parameters on physical models placed in a wind tunnel under controlled conditions. On the other hand, CFD simulations utilise a numerical method to simulate airflow around the panels and estimate the resulting forces, lift, and drag coefficients.

For industrial applications, it is preferable to adhere to standard codes such as the International Building Code (IBC) and American Society of Civil Engineers (ASCE) standard ASCE 7, particularly for the roof-mounted panels. The International Building Code (IBC) serves as a suitable standard for assessing wind loads on structures, including solar panels [13]. Additionally, ASCE 7 provides comprehensive guidelines for determining wind loads and their impacts on solar panel installations [14]. These standards offer insights into wind load requirements based on factors such as building height, location, wind speed, and exposure. However, for certain structures like mobile solar PV systems mounted on trailers (MPSTs), the availability of specific standards and codes may be limited or not directly applicable. This can pose a challenge when it comes to parameter selection in ensuring the structural integrity and resilience of such systems in the face of wind loads.

In cases where standard codes may not exist, it becomes crucial to adopt a rigorous and comprehensive approach to address wind load concerns. This typically involves leveraging engineering principles, expertise, and best practices to design and implement solar PV module systems that can withstand various wind conditions. For such situations, there are two primary approaches that can be employed for predicting wind load effects on solar panels. The first approach involves placing a scaled model of the solar panel in a wind tunnel, where loads are determined using measured parameters on the model [15-17]. The second approach utilises numerical simulation, which has significantly advanced due to the progress made in the information technology (IT) sector. Wind tunnel testing involves constructing a scaled model of the solar panel system and subjecting it to controlled wind conditions [18]. On the other hand, numerical simulations utilise computational models and algorithms to simulate wind flow and its interaction with the solar PV power system [19,20]. Both approaches complement each other, as the results obtained from the wind tunnel testing helps in the validation of the numerical approach. The latter helps with design optimisation and can provide a comprehensive assessment of wind load effects on solar panels.

This paper presents a study that employs both experimental and numerical techniques to investigate the behaviour of the solar PV panels used in the designed MSPT. It focuses on analysing the lift and drag forces exerted on the panels by varying wind velocity, including velocity and direction. The study also evaluates the impacts of wind on the solar PV panels used in the system, considering different panel tilt angles. The Computational Fluid Dynamics (CFD) technique was utilised for the comprehensive investigation and analysis, while the experimental testing was conducted using a scaled-down panel sample with a ratio of 1:15.

3. Methodology

The methodology used in this study was both numerical and experimental. The numerical simulations (using ANSYS-Fluent 2022/R2 software) were conducted on the solar PV panels as well as the entire MSPT. This helped to determine the magnitude and impact of the wind force on the system with different tilt angles and wind velocities. The experimental tests, on the other hand, were used to validate the results obtained from the numerical estimation.

3.1. Experiments

The experimental tests on the panel sample (scaled 1:15) were conducted using an open circuit wind tunnel. The equipment used was the C15-10 subsonic wind tunnel with an incorporated IFD7 interface, which provides connection to a PC on which the software for controlling the experiment is installed. The software interface helps with sensor output logging and controlling the fan speed, as well as performing any required calculations for each test. The wind tunnel has a square test section with nominal dimensions of 150 mm \times 150 mm and 800 mm in length. The base of the wind tunnel's working section

for controlling the experiment is installed. The software interface helps with sensor output logging and controlling the fan speed, as well as performing any required calculations for each test. The wind tunnel has a square test section with nominal dimensions of 150° mm × 150 mm and 800 mm in length. The base of the wind tunnel's working section is removable to allow for the insertion of large or complex models when required. The tunnel is designed with an inlet flow straightener and contraction ratio (9.4:1) to give well developed air flow through the working section. Air is drawn through the working section by a variable speed fan at the discharge end of the tunnel, providing up to 34 m/s air velocity. To prevent personnel and objects from coming into contact with the rotating blades, the fan is equipped with a protective grill on the outside be outside

For this study, the solar panel model was mounted through a circular hatch using quick release clamps (120 mm diameter) in the front wall of the working section. The circular hatch was incorporated with an angular scale, allowing the model to be manually rotated to the desired angle. The transparent working section provided a unique opportunity for flow visualisation around the model.

During the experiment, the fan speed of the wind tunnel was adjusted and kept at a constant speed of 10 m/s, while the angle of attack was increased stepwise from 0 to 35 degrees. The choice of this initial speed was based on the average annual wind speed in the sub-Saharan region, particularly in Nigeria (typically ranging from 2 m/s to 9.5 m/s), where the MSPT is designed to operate. When the desired angle of attack and the velocity have been set, the sensor readings were logged by selecting the start icon on the PC. The jfan speed, pressure, lift coefficient, drag coefficient, as well as other parameters were automatically recorded through the software interface in an Excel sheet on the PC for each dest. The interfaced computer system was used to record and transfer test data, ensuring precision and reliability throughout the experimental procedures. The province taled sold to complete a the operator of the analysis of the analysis of the same of the same of the solution and reliability throughout the experimental procedures. The analysis of the same of t



Figure 1. Solar PV panel sample examined in the wind tunnel.

3.2. Numerical Simulation

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The critical principles for aerodynamic numerical simulations involve judicious considerations of computational domain, flow conditions, geometrical models, boundary conditions, computational schemes, and domain discretisation. These key fundamental

valuating the simulation set-up for the study involving the full-size panel and entire MSPT.

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principles are meticulously discussed in the subsequent subsections, providing a compre-

hensive understanding of the numerical analysis employed in this study. principles are meticulously discussed in the subsequent subsections, providing a compre-hensive understanding of the numerical analysis employed in this study. 3.2.1. Computation Domain and Flow Conditions

Geometrical Models 1. Computation Domain and Flow Conditions

A prototype of the MSPT was constructed and affixed to a trailer, as depicted in Fig-

ure 2. The MSPT configuration configuration configuration of the MSPT was constructed and affixed to be trailer ad depicted in the top and bottom filted at 30 degrees, while the three panels in the middle remain with out tilt. The individual solar PV panels employed in the MSPT measure 2.14 m in without tilt. The individual tilt. The individual solar PV panels employed in the MSPT measure 2.14 m in and 1.08 m in length: Furthermore, upon unfolding the MSPT within 30 hd ssee tiltanglo-degree tilt the panels unveilasighstantial dimensionsubstangth on birstons: a breadth of 63.45, more add a of 3.45 m, height of 1.19 m. and a height of 1.19 m.



Figure 2. Unfolded state of the MSPT switch six parents (top and battom) tilted at Bhdesteest 30 degrees.

Wind Load Wind Load

The velocity of 10 ht/s was selected, as the selected is a thread the designed for the sub-Saharan region, particularly in Nigeria, where the average annual wind speeds typically saharan region, particularly in Nigeria, where the average annual wind speeds typically range from 2 m/s to 9.5 m/s [21,22]. range from 2 m/s to 9.5 m/s [21,22].

Boundary Conditions

Figure 3 depicts a Style presentation of the CHD through the field solar banels and the mid-section of the section of the scaled solar banels and ple (tilt angle, $\alpha = 30$ degrees) simulated using ANSYS Fluent 2022 R2. Appropriate boundary conditions were applied with a symmetry boundary condition at the mid-section of the panels, as illustrated in Figure 3, to simplify the analysis. The panels at different the panels as illustrated in Figure 3, to simplify the analysis. the panels, as illustrated in Figure 3, to simplify the analysis. The panels at different tilting of 10 m/s. angles (-35 to 35 the grees) were isubjected to account antivised nel pairs a following the border and the point of the po provides information an theodim pagious of the panel, models and the loss spanning de-the MSPT, main used in the presedent point wattight, including the file size a parellaised in the MSPT, scaled panel sample, and panel from the literature [23]. Table 1. Model size of the panel (tilted at 30 degrees) and the domain for the present simulations.

	W_P	L_P	L_b	L_{f}	Н	В	Unit
Scaled panel sample (scaled 1:15)	72	145	400	200	150	150	
Full-size panel used in the MSPT	1080	2140	4320	2160	1620	2140	mm
Panel from Lit. [23]	699	349	2794	1397	1034	524	



Figure 3: 3D view of the CFD model (showing the boundary conditions and the meshing system) used to simulate the panel sample examined in the wind turnel.

Jable Ni Model at Model at 30 degrees) and the domain for the present simulations.

	Computation	W_P^{cheme}	L _P	L _b	L_f	Н	В	Unit
Scaled panel sample (sca	aled 1:13 the C	CFD simula	atip <u>n</u> s i	n this gyrren	it stuggy assi	ume ₁ am isot	hermad, vi	scous, and
Full-size panel used in t	he MSPT	le (constar	it densi	ty) fluid The	e incompres	sibl <u>e Re</u> yno Į620	lds ₂₁₄₀ rag	ged Navier
Panel from Lit. [2	3 Stokes (RAN	5) equation	15 for ti	he conservat	1397	and mome 1034	ntum repr	esented by
· · · · ·	-Equations (3)	and (4), and	l descrii	bed in the lite	fature, [24,2	o j govern tni	IS FIOW. THE	governing
	3.2.2. Numer	ical Model	e model tions at	a steady state	e. The gover	study nave ning equatio	been solve ons are disc	a using the retised by a
	Gamputation	abschana	nd the p	oressure-base	d segregated	l solver prop	osed by Cl	norin (1968)
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	<i>ptbsecenyetin</i>	Órin ^r (Fidua	kasestee	ft used for th	theiminitial	vintegration	[26]. In thi	s study, the
	solution met	hod for the	e pressu	are-velocity c	coupling is a	coupled sch	eme with	the second-
	order upwin	d discretisa	ation sc	heme_for+nvo	(læll) in g Ohe	momentum	, turbulent	kinetic (A)
	ergy, and tur	bulent diss	ipation	rate. In addit	ion, the con	vergence of	the solutio	n was mon-
	itored throug	gh the resid	uals for	othe relevant	equations.	When the va	lue of each	n residuaļ js
	between thre	ee to four o	rders o	<i>⊕t</i> magnitude	below its ir	itial value,	the solutio	n is said to
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	densees wheel	lithegeneed	ntune p	notiotherfeater	ationántesie	tuelsyreshie	vtth 40-spet	theispaitial
	coditinates (<i>x, y, z</i>), and	the vel	locity vector	represented	by u . $\partial u/\partial t$	represents	the rate of
	change of vel	ocity with	respect	to time ₀ (t) an	d ∇P denot	es the press	ure gradiei	nt.
	In this s	tudy, the st	tandard	l k-eps ijo ntti	Frb ate h c e∙h	nodel availa	ble in AN	SYS Fluent
		مليم مم ال	1 1		1	f a aalaw DV		7] [T]

In this study, the standard k-epsilon turbulence model available in ANSYS Fillenv was employed to model the aerodynamic behaviour of a solar PV panels [27]. The turbulence kinetic energy, κ , and its duate of dissipation, ε , are obtained from the transport Equations (5) and (6). The standard k-epsilon model is a widely adopted and robust turbul lence model known for accurately representing turbulent flow phenomena. The efficacy of where, $\partial \rho \partial t$ represents the rate of change of density (ρ) with respect to tume (t), and this turbulence model has been highlighted in numerous recent research studies examining V. (ρu) denotes the divergence of the product of density and velocity (ρu) with rethe impact of wind load and patterns on solar PV panels and arrays, making it a frequent spect to spatial coordinates (x, y, z), and the velocity vector represented by u. ou/ot choice for conducting numerical simulations in this context [10,28–30]. By solving the transport equations for turbulent kinetic energy (k) and the dissipation rate of turbulent kinetic energy (ϵ), the model accurately predicts the flow behaviour. Additionally, the standard k-epsilon model was selected due to its versatility and computational efficiency, making it suitable for the simulation of the aerodynamic effects at varying wind velocities and panel tilting angles.

$$\frac{\partial}{\partial t}(\rho k) + \frac{\partial}{\partial x_i}(\rho k u_i) = \frac{\partial}{\partial x_j} \left[(\mu + \frac{\mu_t}{\sigma_k}) \frac{\partial k}{\partial x_j} \right] + G_k + G_b - \rho \varepsilon - Y_M + S_k \tag{5}$$

$$\frac{\partial}{\partial t}(\rho\varepsilon) + \frac{\partial}{\partial x_i}(\rho\varepsilon u_i) = \frac{\partial}{\partial x_j} \left[(\mu + \frac{\mu_t}{\sigma_\varepsilon}) \frac{\partial\varepsilon}{\partial x_j} \right] + C_{1\varepsilon} \frac{\varepsilon}{k} (G_k + C_{3\varepsilon}G_b) - C_{2\varepsilon} \rho \frac{\varepsilon^2}{k} + S_\varepsilon$$
(6)

In Equations (5) and (6),

 G_k = the turbulence kinetic energy due to the mean velocity gradients.

 G_b = the generation turbulence kinetic energy due to buoyancy.

 Y_M = the contributing effects of fluctuating dilatation to the overall dissipation rate. α_k and α_{ε} = the turbulent Prandtl numbers for *k* and ε , respectively.

 S_k and $S\varepsilon$ = user-defined source terms.

$$\mu_t =
ho C_\mu rac{k^2}{arepsilon}$$
, (where $C_\mu = constant$)

The coefficients value for the k-epsilon model were set as presented in Table 2.

Table 2. The coefficients used for the standard k-epsilon model in the present simulation usingANSYS Fluent 2022 R2.

		Empirica	Turbulent Pra	andtl Numbers		
Coefficient	C_{μ}	$C_{1\varepsilon}$	$C_{2\varepsilon}$	$C_{3\varepsilon}$	σ_k	$\sigma_{arepsilon}$
Value	0.09	1.44	1.92	0–1	1	1.3

Domain Discretisation

The computational domain for this study was meshed in the ANSYS Fluent workbench. The unstructured tetrahedral mesh of intermediate mesh density was used for this current study, as shown in Figure 3. The mesh was clustered around the panel surface by the size function tool in the ANSYS Fluent workbench [27] to capture the salient features of the flow.

Mesh Sensitivity

To ensure mesh sensitivity and model accuracy, grid independence tests were conducted for various meshing configurations. Seven different meshes were created by adjusting the number of elements along the panel's edges (N_p) and the ratio of the domain wall's mesh size to the panel's edge mesh size, R_W , as documented in Table 3. For the scaled panel sample tilted at 30 degrees, the number of mesh elements ranges from 224 thousand to 2.7 million.

The predictions of the lift and drag forces on the panel were used as monitoring parameters to establish the grid independence of the predictions reported in Table 3. The results of the studied models demonstrated a narrow range of lift and drag forces, with less than a 2% difference in values. The results indicate that the different levels of computational mesh refinement have little impact on the lift and drag forces. This indicates that the numerical scheme is predicting similar values at all the seven levels of mesh refinement and that any further increase in grid density may not noticeably improve the accuracy of the RANS predictions.

racy of the RANS predictions.

The computational mesh Type 3, with N_p of 100, R_W of 5.6, and N_e of 672 thousand, shows minimum difference in its corresponding drag and lift forces from the mean values *Sustainability***bee4** for **2088** for all the seven case studies. Mesh Type 3 was chosen for this study as it $_{8 \text{ of } 20}$

provides sufficiently grid-independent resolution, given the available computational resources. For the study, the CPU specifications included AMD Ryzen Thread ripper PRO Table 3. Mesh sensitivity of the simulation of the scaled panel sample tilted at 30 degrees. 5975WX 32-Cores processor, a 3.60 GHz processor with 64.0 GB (63.8 GB usable) installed RAM, and a 64-bit opendeling stem, x04-based processor for the system it (00. Addition (N) ally, the aerodynamic simplation of the entire MSPT system 6888 an average of 10 h 4594 the iterations to convergetype 2 1004.2 1,123,048 0.978 0.597 100 5.6 672,426 0.982 0.600 Type 3

Table 3. Mesh sensitivity of the simulation of the scaled panel sample, tilted at 30 degrees.0.600

Model Num	ber Type 5	N_p	¹⁰⁰ R	9.7	N_e 325,716	Lift (N)	Drag (\mathbf{N})
Type 1	6	-100-	⁷² 2.	<u>9</u> 5.6	2,693,877	$-0.972^{0.985}$	0.594
Type 2		100	<u> </u>	2	-1,123,048	-0.978	$-0.597^{0.604}_{0.600}$
Туре 3		100	5.	6 1	672,426	$-0.982_{1.42}$	0.600
Type 4		100	6.	9	478,405	0.980	0.600 3
Type 5		100	9.	7	325,716	0.984	0.600
Туре 6	omp	outation	al mesh J y	фе 3, v	vit h 35, 949 00, F	R _W of 5.6 5and 1	N _e of 670thousand
Type 7	imu 	im diffe 50 or all th	rence in its 5. ne seven c	corres 6 ase stu	ponding drag ar 224,593 dies. Mesh Tyr	id lift forces fro 0.986 <u>e 3 was chose</u>	om the mean values 0,604 n for this study as
	it provides su	fficientl	y grid-ind	epenMe	manesolution, g	iver 0,198 lavail <i>a</i>	ble0c600putational
	resources. For	the stuc	ly, the CP	Janegié	intion% include	d AMD Ryzen	Threpd gipper PRO
	5975WX 32-Co	ores proc	cess or, a 3. Stai	50 GHz	processor with Deviation	64.0 GB (63.8 C	$\frac{B \text{ usable}}{0.003}$ installed

RAM, and a 64-bit operating system, v64-based processor for the system type Additionally, the aerodynamic simulation of the entire MSPT system took an average of 10 h for the

Figure 4 showstetheiovariation endelift and drag forces with different mesh configurations, displaying linear regression lines that capture the trends of the different mesh configurations, as the number of elements in the mesh configuration changes.



Figure 4. Variation of High and Variagio modification of High and Variagio modification of High and Variagio modification of High and the group of the standard of the standar

3.2.3. Numerical Validation

Figure 5 shows the predicted lift and drag forces from the numerical simulation and experimental testing on the scaled solar PV panel sample at a wind velocity of 10 m/s, and varying tilt angles (angle of attack). The results show the drag and lift forces increasing as the angle of attack increases, exhibiting a consistent upward trend. Table 4 shows the results obtained for each of the experiments. Comparing the values of drag and lift forces

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Figures5 Lift had didig coveres control PVP Appensal spite pleale calent 1 at 5) Outs 1/2 m/n dvield or for a first on the please of the ple

Fable 4. Littend dung forses on these sted of pelsemble from experiment and SEP circle stimulations.

AngAngle Attack Attack (Degree)	(Der (Exit)	(E xp.)	(PNA)S (Đứa	g(Nexp.	,)-(N)CTP}	t(\CFD)		jąg _i (CF (N)	Pift Eriff	Error	(%)PBF#gg	É rr or (%)
5 5	0.18	0.18	0.12	0.12	0.26	0.26	0.055	0.055	30.8	30.8	54.2	54.2
¹⁰ 10	0.49	0.49	0.21	0.21	0.52	0.52	0.13	0.13	5.8	5.8	38.1	38.1
15 15	0.64	0.64	0.25	0.25	0.7	0.7	0.23	0.23	8.6	8.6	8.0	8.0
20 20	0.71	0.71	0.32	0.32	0.81	0.81	0.34	0.34	12.3	12.3	5.9	5.9
25 25	0.85	0.85	0.45	0.45	0.88	0.88	0.46	0.46	3.4	3.4	2.2	2.2
30 30	1.07	1.07	0.62	0.62	0.98	0.98	0.6	0.6	8.4	8.4	3.2	3.2
₃₅ 35	1.31	1.31	0.79	0.79	1.18	1.18	0.77	0.77	9.9	9.9	2.5	2.5
	Average	error (Se far an	gles of	attack > 1	5 degre	ee		8.5	8.5	4.5	4.5

Additionally, the accuracy of the current SFD simulation technique was compared anith an unspaked reachber in helin the literature one with the aversity interfightion [23] and the exclusion in the interfightion of the additional sector of the addiation of the analysis are an even the result of the sector of the additional sector of the additi

4. Results

This section presents the results of predicted numerical simulations of the flow interaction and behaviour around a single solar panel used in the MSPT and the entire MSPT. The unscaled panel shares a similar aspect ratio of 2 to the panels used in the MSPT, but has different dimensions, as specified in Table 1 (with a width of 699 mm and a length of 340 mm). The drag and lift forces obtained from the current simulation of the tilted panel

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from the current CFD simulation. As depicted in Figure 6, the leading edge of the p facing the incoming airflow experienced positive pressure, while the trailing edge o panel encountered negative pressure, indicating the suction side.

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⁸ Figure 7a,b depict, respectively, the vector of flow velocity in the x direction and vectors of flow velocity magnitude (inlet to outlet), showing the detailed flow behav around the panel from the simulation as in Figure 6. Figure 7 illustrates the flow sep tion occurring interest and see of a the panel accompanied by high during inf and the set of a the second degree of the second degree of the second degree and the second degree and the second degree of the second degree and the second degree of the second degree of the second degree and the second degree and the second degree of the second degree of the second degree and the second degree and the second degree and the second degree and the second degree of the second degree and the

Static Pressure [Pa]



Figure Gistratic pressure contour at the microsciction to fulles fulles ize yokane Puep and euter in the N tilted at 190 dtg elegrees.

Figure 7a,b depict, respectively, the vector of flow velocity in the x direction and the vectors of flow velocity magnitude (inlet to outlet), showing the detailed flow behaviour around the panel from the simulation as in Figure 6. Figure 7 illustrates the flow separation occurring at the edges of the panel, accompanied by high turbulence and the generation of vortices at the back of the panel, one near the top edge and the second further away from the bottom edge in the wake region of the panel. The location of the vortices was the low-pressure region behind the panel in Figure 6. The separated flow at the top and bottom edges of the panel resulted in an increase in velocity at the edges, reaching up to 15 m/s. Additionally, the vortices at the back of the panel induced a recirculation flow pattern on the opposite side of the inlet flow with an average velocity of 5.5 m/s, thus indicating suction effects at the back of the panel.

Figure 8a shows a 3D representation of the flow structure (streamlines of velocity magnitude) around the MPST, which provides further details about the flow structure around the MPST discussed in Figure 7. The result in Figure 8 shows the minimum flow velocity occurring at the back of the panels in the form of recirculating vortex structures, while the maximum flow velocity of 16 m/s was observed at the edges of the panel. The features show the solar panels retarding the incoming flow velocity as the flow passes over the panels. The vortex shedding at the back of the panel occurred due to the flow separation around the structures. This recirculation is more visible and pronounced in Figure 8b.

4.1. Effects of Wind Velocity

Figure 9a,b illustrate the variations in drag and lift forces, as well as the corresponding drag and lift coefficients, for the full-size panel used in the MSPT. These variations were observed under different wind velocities and at tilt angles of 15 degrees and 30 degrees.

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(b)

Figure F7g(a) NextVerstors of olan wellocity in the direction (indetter). (b) flet) of (b) NextVerstors of flow ver magnitude in the discretion of the full isize iso BYPP version in the second of the full isize iso BYPP version of the full isize iso BYPP version in the second of the full is a second of the se

Figure 8a shows a 3D representation of the flow structure (streamlines of vel magnitude) around the MPST, which provides further details about the flow stru around the MPST discussed in Figure 7. The result in Figure 8 shows the minimum velocity occurring at the back of the panels in the form of recirculating vortex struct while the maximum flow velocity of 16 m/s was observed at the edges of the panel features show the solar panels retarding the incoming flow velocity as the flow p over the panels. The vortex shedding at the back of the panel occurred due to the



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Figure 8. (a) 3D Figures in the of most structure of the structure (streamlines of velocity magnitude) the und the MSPT (tilted at 30 degrees), (b) 2D flow structure (streamlines of velocity magnitude).



well as the correspond-MSPT. These variations of 15 degrees and 30 de-

gles, demonstrating that d velocities rise. For incposed to a 10 m/s wind, . Doubling the wind vee) and 377 N (lift force). mained nearly constant and 0.66, respectively. ed to a 10 m/s wind, the easing the wind velocity ag force) and 483 N (lift ted panel remained conespectively.



4.2. Effects of Tilt Angle

Figure 10 illustrates the variation of drag and lift forces, as well as the corresponding lift and drag coefficients on the full-size panel with an aspect ratio of two (width: 1083 mm, length: 2140 mm) at different tilt angles (-35 to +35 degrees) and with 10 m/s wind velocity. It is evident from Figure 10 that the trend of drag and lift forces closely follows the trend of the coefficients.

In agreement with previous results (representing a panel with a different size and

The figures illustrate a consistent trend across different tilt angles, demonstrating that drag and lift forces increase with a quadratic relationship as wind velocities rise. For instance, in Figure 9a, when the panel was tilted at 15 degrees and exposed to a 10 m/s wind, the drag and lift forces measured 32.8 N and 94.2 N, respectively. Doubling the wind velocity to 20 m/s led to an approximate increase to 128 N (drag force) and 377 N (lift force). Notably, the values of C_L and C_D for the 15-degree tilted panel remained nearly constant across different wind velocities, with approximate values of 0.23 and 0.66, respectively.

In Figure 9b, with the panel tilted at 30 degrees and subjected to a 10 m/s wind, the drag and lift forces were 76 N and 121 N, respectively. Upon increasing the wind velocity to 20 m/s, the corresponding forces were approximately 303 N (drag force) and 483 N (lift force). Remarkably, the values of C_L and C_D for the 30-degree tilted panel remained consistent, unaffected by changes in wind velocity, at 0.53 and 0.85, respectively.

4.2. Effects of Tilt Angle

Figure 10 illustrates the variation of drag and lift forces, as well as the corresponding lift and drag coefficients on the full-size panel with an aspect ratio of two (width: 1083 mm, length: 2140 mm) at different tilt angles (-35 to +35 degrees) and with 10 m/s Wind Velocity. It is evident from Figure 10 that the trend of drag and lift forces closely follows the trend of the coefficients.



Figure 10. Drag and **Figure 10.** Drag and

In agreement with previous results (representing a panel with a different size and 4.3. Effect of Panel Seate Fraction, 0.5), it was apparent from Figure 10 that the coefficients C_L and C_D were not influenced by the direction of the tilt angle. Also, comparing the results from Figure 10 and To investigate the impact of panel size on the aerodynamic coefficients of solar panels, a further CFD study was conducted on panels with seating scaling factors in pesifigation ically, simulations of the effective of sections of the interaction between the different parameters and the scaling factors ranging from 0.1 to 1. The results of the interaction between the different parameters and the scaling factor decreases and the panel size reduces, the drag and lift forces decrease significantly. This decrease may be attributed to the reduced surface area of the panel that was exposed to the wind. However, the results show that the corresponding coefficients, C_L and C_D , did not exhibit pronounced changes with varying panel size. These coefficients, C_L and C_D , did not exhibit pronounced changes with varying panel size.

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cients, L_L and L_D .

4.3. Effect of Panel Scale Factor

Sustainability 2010 A provestigate the impact of panel size on the aerodynamic coefficients of solar₁ $_{PO20}^{+}$ panels, a further CFD study was conducted on panels with various scaling factors. Specifically, simulations were conducted on the panel with dimensions of 1083 mm in width and 2140 mm in length, tilted at 30 degrees and exposed to a wind velocity of 10 m/s, with the scaling factors ranging from the impact of panel size on the aerodynamic coefficients of solar PV the scaling factors ranging from the impact of panel size on the aerodynamic coefficients of solar PV the scaling factors ranging from the panel with dimensions in disatenthat wanthat scaling factor decreases and the panel size of the forces and states to califor the forces and from the panel size on the aerodynamic coefficients of solar PV trends of the forces and from the panel of panel size on the aerodynamic coefficients of solar PV panels, a utrike CFD soudy was conducted on panels with various scaling factors. Specifitrends of the forces and from the panel size of the different scaling factors scaling factor decreases the prime of the tore of the tore of the tore of the panel size on the forces and coefficients for different scaling factors and indicate that as the scaling was exposed to the forces and coefficients for different scaling factors and indicate that as the scaling was exposed to the forces and coefficients for different scaling factors and indicate that as the scaling was exposed to the forces and coefficients for different scaling factors and indicate that as the scaling was exposed to the forces and coefficients for different scaling factors and indicate that as the scaling was exposed to the forces and coefficients for different scaling factors and indicate that as the scaling was exposed to the forces and coefficients for different scaling factors and indicate that as the scaling was exposed to the forces and coefficients for different scaling factors and indicate that as the scaling was exposed to the forces and





4.4. Effects of Panel Aspect Ratio

Figure 12 presents the variation of drag and lift forces and their corresponding coefficients for panels with different aspect ratios between 0.5 and 3. These panels, tilted at 30 degrees and exposed to a wind velocity of 10 m/s, had a fixed width of 1083 mm, while the length ranged from 541 mm to 3249 mm. The findings from the results shown in Figure 12 reveal that panels with aspect ratios less than 1.5 exhibited higher values of C_L and C_D compared to panels with higher aspect ratios. For example, the values of C_L and C_D were 1.15 and 0.69, respectively, for the panel with an aspect ratio of 0.5. The results also show a steady increase in lift and drag coefficients with aspect ratios greater than 1.5, with the values of C_L and C_D ranging from 0.51–0.58 and 0.81–0.93, respectively. The observed differences in the values of C_L and C_D can be attributed to the influence of the panel aspect ratio on the aerodynamics of the panel.

4.5. Aerodynamics of MSPT

A full-scale representation of the MSPT as illustrated in Figure 2 for the fabricated prototype, consisting of nine panels tilted at a constant angle of 30 degrees, was modelled following the previous simulation set-up outlined in Section 3. However, the symmetry boundary condition was not applied to the MSPT due to inherent asymmetries in the middle of the system. The dimensions of the MSPT, denoted as W_P and L_P , were 3612 mm and 6614 mm, respectively. Additionally, the dimensions of the corresponding simula-





0.80

4.5. Aerodynamics of MSPT

3A full-scale representation of the MSPT as illustrated in Figure 2 for the fabricated prototype, consisting of nine panels tilted at a constant angle of 30 degrees, was modelled following the previous simulation set-up outlined in Section 3. However, the symmetry boundary condition was not applied to the MSPT due to inherent asymmetries in the middle of the system. The dimensions of the MSPT, denoted as W_P and $L_P^{0,0}$ were 3612 mm and 6614 mm, respectively. Additionally, the dimensions of the corresponding simulation domain, encompassing L_b , L_f , H_i and B_i were 15,000 mm, 7250 mm, 5418 mm, and 13,228 mm respectively. Figure 13, provides a 3D, representation of the computational domain between the mid-





A wind velocity of 10 m/s was employed) as detailed in Section 3.2.1. However, to offer a more comprehensive understanding of the aerodynamic performance of the MSPT, more rigorous parametric studies were explored with wind velocities of 15 m/s and 20 m/s.

Figure 14 shows the static pressure contour of the panels in the entire MSPT (tilted at 30 degrees). The results in Figure 14 show that the bottom of the tilted panels experienced the maximum static pressure as the air flows over the entire MSPT, while the middle panels with no tilt angle exhibited minimum pressure.

Figure 15 shows the velocity vectors of flow in the x direction at the mid-section of the MSPT. As observed in Figure 15, flow separation led to a higher velocity magnitude, **Figahing.** Op to 2011/16 COD parted, to the injustre flow and input to a higher velocity magnitude, **Still States and Still States** (to the parted with the parted states) imum velocity of 6.2 m/s in the opposite direction to the inlet flow.

A comparison between Figures 7 and 15 reveals that the magnitude of the suction flow velocity around the MSPT was almost 62% higher compared to the individual panel. Also, comparing the results from Figures 6 and 14, the suction pressure occurring at the back of panels in MSPT was approximately 22% greater than that of a single panel. These observed differences may be attributed to the presence of gaps between panels in the MSPT, leading to an increase in turbulence.

more rigorous parametric studies were explored with wind velocities of 15 m/s and 20 m/s.

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Figure 14 shows the static pressure contour of the panels in the entire MSPT (tilted at 30 degrees). The results in Figure 14 show that the bottom of the tilted panels experienced the maximum static pressure as the air flows over the entire MSPT, while the middle panels are 16 of 20 els with no tilt angle exhibited minimum pressure.







Figure 15. Velouity veeto of flow in the dictination (in link of the the suid ion of the MSPIMSPT.

5: Discussion

A comparative analysis was undertaken to obstantiate the fraction of the first the analysis was undertaken to obstantiate the fraction of the first obstantiate the fraction of the first obstantiate the first obstantiate

ness of the results obtained in this study. Figure 16 also shows that the coefficients exhibit nearly identical values for each tilt angle, indicating that the aerodynamic performance of the panels remains consistent regardless of the wind's direction relative to the panel orientation. This suggests that the aerodynamic coefficients are not affected by the facing direction of the wind (from the front side or back side), whether it is approaching from the front side or the back side, as also reported by Giorges et al. (2013) [23]. This finding isginneatantoand is in the front side or back side of the side of the side of the side of the back side, as also reported by Giorges et al. (2013) [23]. This finding isginneatantoand is in the front side of the side of the side of the side of the back side, as also reported by Giorges et al. (2013) [23]. This finding isginneatantoand is in the front side of the side of the side of the side of the back side, as also reported by Giorges et al. (2013) [23]. This finding isginneatantoand is in the side of the back side, as also reported by Giorges et al. (2013) [23]. This finding isginneatantoand is in the side of t



Figure 16. C_L and C_D for the unscaled panel (with a width of 699 mm and a length of 340 mm) at **Figure 16** in G_D for the unscaled panel (with a width of 2599 mm and a length of 340 mm) at different tilt angles from the present study and the literature [23,31].

Figure 17 illustrates the drag and lift forces on the entire MSPT and the single panel Figure 17 illustrates the drag and lift forces on the entire MSPT and the single panel under varying wind velocities. An increase in the drag and lift forces as the velocity inunder varying wind velocities. An increase in the drag and lift forces as the velocity inunder varying wind velocities. An increase in the drag and lift forces as the velocity inunder varying wind velocities. An increase in the drag and lift forces as the velocity increases from 10 m/s to 20 m/s for both the single panel and the entire MSPT is observed increases from 10 m/s to 20 m/s for both the single panel and the entire MSPT is observed. For instances, the drag and lift forces on the entire MSPT under a wind velocity of 10 m/s were found to be 511 N and 941 N, respectively, whereas the drag and lift forces on the entire MSPT at a wind velocity of 20 m/s were found to be 2038 N and 3747 N, respectively, entire MSPT at a wind velocity of 20 m/s were found to be 2038 N and 3747 N, respectively, entire MSPT at a wind velocity of 20 m/s were found to be 2038 N and 3747 N, respectively. The data analysis indicates that the drag and lift forces at a wind velocity of 20 m/s are spectively. The data analysis indicates that the drag and lift forces at a wind velocity of 20 m/s are spectively. The data analysis indicates that the drag and lift forces at a wind velocity of 20 m/s are spectively. more than four times greater than the corresponding forces at a wind velocity of 10 m/s.



Figure 17. Drag and lift forces acting on the single panel and the entire MSPT system under different wind velocities.

Additionally, Figure 1¹⁷shows the day and differences the third way for the provident of the provident of

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Equations (1) and (2) and taking the total area of the nine panels in the MSPT as 20.85 m^2 , and the air density to be 1.22 kg/m^3 , the drag and lift coefficients for the entire MSPT were determined to be 0.40 and 0.74, respectively, for all the three investigated wind velocities. It is important to note that these coefficients remained consistent and were not influenced by changes in wind velocity. As discussed in Section 4.1, it is noteworthy that for a single panel with a tilt angle of 30 degrees, the drag and lift coefficients were determined to be 0.53 and

0.85, respectively. This implies that simple superposition assumptions, such as considering nine or six times the forces on a single panel (accounting for the nine panels, with six tilted at 30 degrees), does not provide sufficient accuracy for determining the aerodynamic forces on the entire MSPT.

6. Conclusions

Mobile solar PV systems are becoming increasingly recognised as crucial solutions for clean and renewable energy, especially in off-grid areas. This study employed numerical and experimental techniques to scrutinise the aerodynamic characteristics of an innovatively designed MSPT system, featuring nine large solar PV panels.

The focus of this study was an average wind velocity of 10 m/s, reflective of Sub-Saharan Africa, specifically Nigeria's annual wind speeds. The investigation explored the impacts of wind velocity and tilt angles on drag and lift forces. The primary comparative parameters for aerodynamic analysis were the lift and drag coefficients (C_L and C_D). Employing ANSYS-Fluent 2022/R2 software, the simulations for a single solar PV panel were validated through experimental tests on a 1:15 scaled-down model. Additionally, the study validated the numerical approach by comparing the results with documented cases of a solar PV panel found in the literature.

Moreover, this study explored the impact of several factors on the aerodynamics of solar PV panels. Primary considerations encompassed tilt angles and wind velocities. Furthermore, recognising gaps in the existing literature, the investigation was extended to the impact of scaling size and the aspect ratio of the panels on the aerodynamics and flow behaviour around the panels.

The results across all studied scenarios indicated that aerodynamic coefficients remained unaffected by wind direction, whether the wind approached the solar panel from the front or back side. The CFD analysis on a single full-size panel in the MSPT revealed a nonlinear increase in drag and lift forces with higher wind velocities. Scaling down the panels showed a significant decrease in forces due to reduced wind exposure, following a power-law relationship. However, lift and drag coefficients exhibited minimal changes with varying panel size. Panels with aspect ratios below 1.5 demonstrated higher C_L and C_D values compared to those with higher aspect ratios.

Examining the overall wind force on the entire MSPT accounted for combined wind flow and system geometry effects. The CFD results indicated increased turbulence caused by gaps between panels, leading to almost 62% higher suction flow velocity and 22% higher suction pressure compared to a single panel. Furthermore, drag and lift forces on the entire MSPT were approximately 6.7 and 7.8 times greater than those on a single panel with the same 30-degree tilt angle. The corresponding drag and lift coefficients for the entire MSPT were found to be 0.40 and 0.74, respectively, while for a single panel tilted at 30 degrees, the values of C_L and C_D were 0.53 and 0.85, respectively. It is clear from this study that simple superposition assumptions, considering the forces on a single panel, cannot accurately predict the aerodynamic forces on the entire MSPT. Therefore, a further aerodynamic investigation is found to be necessary to determine the wind effects on the entire system more accurately. Therefore, the author would like to recommend further aerodynamic investigations to precisely determine the wind effects on the entire system.

In conclusion, this study provided significant insights into the aerodynamic behaviour of mobile solar PV systems, facilitating a comprehensive understanding of their performance under diverse wind conditions. These findings would be vital for optimising the design and operation of such systems, ultimately enhancing their efficiency and reliability, especially in Sub-Saharan Africa. The integration of CFD simulations and experimental validation contributed to the knowledge in the field, opening avenues for further advancements in mobile solar PV technology.

Author Contributions: Conceptualisation, A.E.M., F.T.-M. and D.S.A.; methodology, A.E.M. and D.S.A.; investigation, A.E.M. and D.S.A.; resources, A.E.M., D.S.A. and F.T.-M.; data curation, A.E.M., F.T.-M., D.S.A., N.N.E., J.W., D.N. and Z.M.; writing—original draft preparation, A.E.M., F.T.-M. and D.S.A.; writing—review and editing, A.E.M., F.T.-M., D.S.A., N.N.E., J.W., D.N. and Z.M.; visualisation, A.E.M., F.T.-M., D.S.A., N.N.E., J.W., D.N. and Z.M.; project administration, F.T.-M. and D.S.A.; funding acquisition, F.T.-M. and D.S.A. All authors have read and agreed to the published version of the manuscript.

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