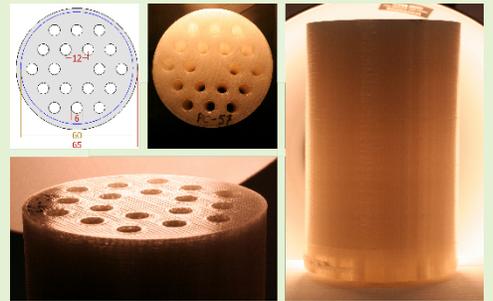


# Bragg gratings inscribed in solid-core microstructured single-mode polymer optical fiber drawn from a 3D-printed polycarbonate preform

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**Abstract**—This paper reports the first microstructured solid-core fiber drawn from a 3D-printed preform and the first fiber Bragg gratings inscribed in a fiber of this type. The presented fiber is made of polycarbonate and displays single-mode behavior. The fiber attenuation was the lowest reported so far for a POF drawn from a 3D-printed preform across a broad range of wavelengths. In addition, extensive fiber characterization results are presented and discussed including: fiber attenuation, mode simulations, dynamic thermomechanical analysis and thermo-optic coefficient. Fiber Bragg gratings are successfully inscribed in the produced fiber using three different lasers: a continuous wave helium-cadmium laser, a pulsed femtosecond frequency doubled ytterbium laser and ultra-violet nanosecond krypton fluoride laser. Mechanical testing of the fiber showed that the 3D printing approach did not introduce any unexpected or undesirable characteristics.

**Index Terms**—Fiber optics sensors, fiber Bragg gratings, microstructured fibers, fiber characterization, additive layer manufacturing, 3D printing, fused deposition modeling.



## I. Introduction

THE group of techniques commonly described as 3D printing or additive layer manufacturing (ALM) have been revolutionizing the field of manufacturing and rapid

prototyping in recent years and are increasingly envisioned as the manufacturing techniques of the future [1, 2]. Among their advantages lie the rapidly increasing quality of fabricated parts; ease and speed of customizability; decreasing price; ability to

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reproduce complex shapes that are difficult to manufacture otherwise; and a growing number of printing materials which offer ever more potential applications.

Fused deposition modelling (FDM) is the most popular 3D printing approach. It starts with melting thermoplastic filament in a hot printing nozzle. Subsequently, the resulting semi-liquid material is pushed through the printing nozzle and deposited in the desired location for the model being printed. After leaving the printing nozzle, the molten material cools down and solidifies to build up the manufactured object layer by layer.

In theory, FDM should be intrinsically well-suited for manufacturing fiber preforms for the production of polymer optical fibers (POFs), as the basic material requirement for both POF preforms and FDM filament is thermoplasticity. The biggest advantage of using the FDM technique for fabricating POF preforms is ease of reproducing complex shapes that would be very difficult to manufacture with other methods such as drilling techniques. Moreover, FDM offers the broadest printing material range of all ALM techniques. In addition, printing polymers can be mixed within one preform printout, which allows for the potential tuning of the material properties (including refractive index, transparency, chemical composition, flexibility, biodegradability, coefficient of thermal expansion, affinity to water etc.).

The qualities of FDM applied to POF manufacturing (predominantly for visible and infrared range) have already been recognized, leading to several interesting publications in the recent years. A step-index fiber was drawn from a preform made of two polymers, printed on a dual-head 3D printer [3]. The core of this fiber was made of styrene-butadiene copolymer and polystyrene (SBP) and cladding of modified polyethylene terephthalate glycol (PETG). In [4], cores of various shapes (made of polycarbonate, PC) and complementary claddings (made of acrylonitrile butadiene styrene, ABS) were printed separately and assembled together to form preforms, which were subsequently drawn. This demonstrated the ease at which 3D-printed shapes and hence fiber properties could be customized. Air-structured fiber drawn from a 3D-printed preform made of SBP was presented in [5]. Hollow-core 3D-printed POF preforms have also been reported. The first one proposed was made of poly(methyl methacrylate) (PMMA) and drawn to cane stage [6], and the first HC POF drawn from 3D-printed preform was made of ABS [7]. This was followed by a publication on mid-IR HC microstructured POF (mPOF) drawn from 3D-printed preform made of PETG [8]. Coreless POFs drawn directly from 3D printer nozzle were reported in [9], in which preform stage in POF manufacturing was omitted. In addition, larger THz waveguides have been directly 3D-printed at the size required for use [10, 11, 12].

The aim of this work is to advance further POF manufacturing research with regards to the 3D printing of preforms and demonstrate the feasibility of fabricating single-mode microstructured solid-core fibers in order to further understand some of the limiting factors associated with the technology. This opens up the opportunity for complex perform designs, comprising of single or multiple materials, that are difficult to realize with conventional techniques to be directly printed in a single-stage process. For example, with hollow-core fibers this could include hollow-core negative curvature

fibers that contain nested elements, ellipses or half-ellipses, which have been modelled but are not feasible to make with conventional fabrication methods. The work also aims to produce a fiber suitable for the inscription of fiber Bragg gratings at 870nm and 1560nm. Extensive fiber characterization results are presented in order to understand the properties of the fiber.

The prototype preform was printed using PC, which has previously been used to successfully fabricate POFs using the drill-and-draw technique [13, 14]. PC was chosen due to the highest glass transition temperature (145°C) and highest operational temperature (125°C) of all POF materials reported so far [13, 14]. For reference, the highest reported operational temperature of Topas 5013S-04 is 110°C [15], of Zeonex 480R is 100°C [16] and of PMMA (the most popular POF material) is 92°C [17].

Dynamic mechanical analysis (DMA) was used to evaluate the mechanical characteristic of the fiber. DMA is a well-established method for polymer characterization used in different fields including industrial [18], automotive [19], aircraft [20] and biological applications [21]. In this technique, a sample is fixed at one end, and an oscillatory tensile load is applied on the other. This results in performing sequential strain cycles with controlled frequency and displacement. Moreover, the method can also involve temperature variation. This then allows evaluating dependency of Young's modulus of sample on all the mentioned parameters (strain, temperature and strain cycle frequency). Such analysis has previously been used to characterize POFs of various structures and made of different materials, including PMMA microstructured polymer optical fibers (mPOFs) [22], PMMA step-index fibers [23], mPOFs of different materials (Topas 5013S-04, Zeonex 480R, polycarbonate) [24], and CYTOP graded-index fibers [25].

Despite still being considered an emerging family of optical fiber sensors (OFSs), polymer optical fiber Bragg gratings (POFBGs) [26] have been attracting growing attention. They share common advantages of OFSs, which include: being lightweight, small and potentially low cost; ease of integration with existing fiber optic networks; the potential for use in many applications for which electronic sensors are intrinsically unsuitable such as in harsh, flammable and explosive environments; ease of multiplexing; immunity to electromagnetic interference; remote interrogation ability. POFBGs are often contrasted with silica optical fiber Bragg gratings (silica FBGs or SOFBGs) in order to highlight their particular qualities. These entail much lower elastic modulus (3.3 GPa for POFBGs made of PMMA compared to 73 GPa for SOFBGs); much higher yield strength resulting in greater strain sensing range; biocompatibility and biodegradability; flexibility in modification of chemical composition and structure, allowing fine tuning of the fiber material properties e.g. to make it sensitive or insensitive to water [26].

POFBGs are capable of sensing various parameters e.g.: strain [27, 28, 29, 30], bend [31], other mechanical deformation [32, 33], temperature [27, 28, 34], humidity [14, 35, 36], pressure [33, 37, 38, 39], refractive index [40] and acceleration [41]. The most used material for fabricating POFBGs is PMMA [27, 28]. However, POFBGs have been successfully manufactured in several other polymer types: polycarbonate

(PC) [14], Topas [42, 43], Zeonex [16] and CYTOP [31]. The application of POFBGs has been demonstrated in a number of fields including: medicine [44, 45, 46], biochemistry [40], aviation [47, 48], dosimetry [49], cultural heritage preservation [50] and integration with textiles [50].

Inscribing FBGs in POFs drawn from 3D-printed preforms is potentially very attractive in terms of being able to further broaden application range of POFBGs through the use of bespoke fiber designs for particular applications.

The paper commences with describing research leading to optimization of transparency of 3D printouts made of PC. Based on this work, a POF preform is printed and characterized. Secondly, preform post-processing and fiber drawing is presented. The resulting POF is characterized, which include cleaving and light coupling to core, modelling of the propagation, cut-back attenuation measurements and dynamic thermomechanical analysis. Finally, FBGs inscribed with different lasers and techniques are shown and characterized, and the thermo-optic coefficient of the fiber is calculated.

## II. PREFORM AND FIBER FABRICATION

### A. Transparency Optimization of 3D Printouts

All preforms described in this paper were produced using an UP! Plus 3D printer (model 3DP144A) and its proprietary slicer software with the FixUp3D extension. Commercially available PC filament of 1.75mm diameter was used (manufactured by Dr3DFilament).

Before 3D-printing the preform, the printing parameters had to be optimized so as to achieve maximal transparency of the printed structure. The biggest challenge lies in air being trapped between thread lines during the fabrication process, which results in light scattering defects being created in preform. Transparency optimization relies then on ensuring that adjacent thread lines melt together well with no air voids between them.

To optimize the transparency a series of thin (4 mm) circular samples of diameter equal to the preform (65 mm) were printed. The key printing parameters were varied, including the printing speed (speed of movement of printing nozzle), filament feed rate and nozzle temperature. The values for layer thickness, nozzle diameter and printing bed temperature were kept constant. After printing, the samples were polished from top, bottom and side, and the transparency was evaluated visually.

The printing speed was varied from 5 to 20 mm/s. Generally, lower printing speeds resulted in better transparency (see Fig. 1) probably due to heat transfer from nozzle to printout. The higher the temperature, the better the infill threads melt together, and the lower printing speed allows a given region of printout to absorb more heat from nozzle as it approaches before the material is deposited. Such a mechanism would also explain the difference in transparency between the samples printed on heated (105°C) and non-heated printing bed (compare Fig. 1 (c) and (d)).

The layer thickness was set to 150  $\mu\text{m}$ , and diameter of the nozzle used was 300  $\mu\text{m}$ . It had been found previously that the optimal printing temperature from the viewpoint of transparency for PMMA was 280-290°C [6]. Due to its higher melting point, PC would ideally be printed at higher nozzle temperatures than PMMA. Consequently, the nozzle

temperature was fixed at 300°C, which was the maximum settable value for the 3D printer used.

Finally, the material infill was optimized by varying line separation and filament feed rate (software parameters: hatch width 0.28mm, feed scale x0.75 and hatch scale x0.99). Unfortunately, due to the control software, it is not possible to give an absolute value for the filament feed speed. Both too high and too low quantities of material infill were found to decrease transparency. Too low values resulted in volume of printout not being fully filled in. Too high feed rates made excessive material aggregate on printout walls which in turn lowered the printing accuracy and hindered operation of the printer.

### B. Fabrication of the Preform

The rationale behind the initial preform design was to make it as easy as possible to be reliably reproduced by the printer. Consequently, a two-ring structure in a hexagonal arrangement was printed (top view shown in Fig. 2(a)), which has previously been shown to guide light [51]. The hole diameter ( $d$ ) was set to 6 mm and the hole pitch ( $\Lambda$ , distance between centers of adjacent holes) to 12 mm. The preform diameter and height were 65 mm and 105 mm, respectively. Before drawing, the preforms were machined down to 60 mm and 100 mm, respectively.

The preform (Fig. 2) was printed at a speed of 5 mm/s (other settings were the same as in Section II.-A), taking ~323 hours in total and using 291.7 g of material. The printout warped a little during printing (the sides of the raft detached from the printing bed), resulting in a slightly rounded bottom surface. In the case of this preform, warping was not an issue, because it did not distort the designed geometry above the bottom surface. The quality of the top and bottom sections is not overly important, since they are sacrificed during fiber drawing stage.

### C. Preform Post-Processing and Fiber Drawing

The finished preform was annealed in a conventional oven at 130°C for 4 weeks to remove air bubbles trapped in preforms and hence increases transparency [3, 5]. It was then machined in order to remove surface roughness and make it suitable for cane drawing using the polymer draw tower. The resulting length and outer diameter were 100 mm and 60 mm, respectively and an outer surface roughness of ~3-4  $\mu\text{m}$ . The preform was then further annealed prior to drawing for 4 more weeks under the same conditions.

The fiber was produced using the two-step heat and draw method [51] where first, the preform was drawn into intermediate canes at a temperature of 180°C. The canes were 6 mm in diameter and 50 cm in length. Each cane was subsequently inserted in a PC sleeving tube. The resulting composite canes were annealed for 10 days at 130°C to ensure that any bubbles trapped in the sleeving tube and cane were removed. Then, one of the composite canes was drawn to fiber at a temperature of 185°C. In the remaining part of the text, the resulting fiber is referred to as 3D PC mPOF.

## III. FIBER CHARACTERIZATION

### A. Fiber cleaving and light coupling to core

Fig. 3(a) shows a microscopic image of a cleaved facet of the 3D PC mPOF. The darker inner region corresponds to the

perform, and the lighter outer one corresponds to the sleeving tube. In the magnified image (Fig. 3(b)), the core is clearly visible. The literature reports the optimal cleaving temperature for a PC mPOF fiber to be 80°C [13, 14]. In this instance a temperature of 75°C yielding the best cleave quality for the temperatures tested where both the fiber and the blade were heated. However, this temperature was possibly a little too high for this fiber as can be inferred from the slight ellipticity of the microstructure holes [52]. In this research, fiber cleaving was performed manually using a blade and a hot plate, which only gave limited control over the process. Further optimization of cleaving temperature was not undertaken as the cleave quality was found enough for light coupling. Such optimization often involves specialized cleaving apparatus and is a separate research question [52, 53, 54] beyond the scope of this work.

Red light from a fiber-coupled semiconductor laser (wavelength ~638 nm, optical power ~2.5-5.0 mW) was butt-coupled into the fiber (fiber length = 38.1 mm) and projected onto a screen (experimental setup in Fig. 4) to observe the clear core guidance, shown in Fig. 3(c). Core guidance was visible by eye over fiber lengths of ~10 cm. For this wavelength of light, only cladding guidance from the outer tube was observed at longer fiber lengths (see Fig. 3(d)).

### B. Simulation of the Confinement Loss and Modality of the Fiber

For a microstructured fiber to display endlessly single-mode behavior, the  $d/\Lambda$  ratio should be less than 0.406 [55]. Here the chosen dimensions of hole diameter and pitch of the preform would result in  $d/\Lambda = 0.5$ . After drawing, the geometry of the fiber changed and the average diameter of the holes in the inner ring ( $d_{\text{inner}}$ ) was visibly higher than of those in the outer ring ( $d_{\text{outer}}$ ). The measured geometry was:  $d_{\text{inner}} = 2.7 \mu\text{m}$ ,  $d_{\text{outer}} = 1.45 \mu\text{m}$ ,  $\Lambda = 5.4 \mu\text{m}$  (see Fig. 3(b)).

For non-endlessly single-mode mPOF, there is no strict criteria to differentiate single-mode and multi-mode behavior. As opposed to step index fibers, all modes in microstructured optical fibers have some non-zero confinement loss due to finite number of microstructure rings. However, a large enough number of rings (often 8) ensures that the confinement loss is negligible compared to other loss mechanisms [51, 56]. Hence, fiber modality is often designated based on acceptable value of loss for a particular application, and in the case of high loss for second mode, fiber can be considered “essentially” single-mode. A useful approximation of the cut-off wavelength based on fiber geometrical parameters was presented in [55]. There, the cut-off wavelength was defined as the transition between delocalized single mode (the second mode not being localized in core) and localized multimode behavior (second and possibly higher order modes localized in core). However, this cut-off wavelength was defined for an ideal silica fiber made of 8 rings of even holes.

In order to assess the confinement loss and modality of the fiber, COMSOL (v. 4.3) [57] and CUDOS (v. 2) [58, 59, 60] software packages were used to run simulations on the described geometry. They gave very similar results regarding real part of effective refractive index ( $\text{Re}(n_{\text{eff}})$ ) and modal patterns, while CUDOS seemed to provide higher precision of imaginary part of  $n_{\text{eff}}$  ( $\text{Im}(n_{\text{eff}})$ ). Simulations were performed for

the two wavelengths (870 nm and 1550 nm, see Fig. 5 and Table I) at which FBGs were inscribed (Section IV). Material refractive indexes used were 1.577 and 1.567, respectively [13]. Simulations at 1550 nm resulted in virtually the same plots as at 870 nm (Fig. 5).

Modes of microstructured fibers follow patterns of step-index modes only approximately. The step-index fiber naming convention (HE, EH, TM, TE) does not work for more complex mode shapes and so designation of modes based on symmetries is used instead [61]. The symmetry group of 3D PC mPOF is  $C_{6v}$ . This gives 8 classes of modes: 4 non-degenerate ones (1, 2, 7, 8) and two pairs of two-fold degenerate classes (3 and 4, 5 and 6).

From simulation, the confinement loss of the second mode was found to be much larger than the fundamental mode at both 870 nm and 1550 nm (220 times and 122 times, respectively). The second least lossy mode at both wavelengths (equivalents of  $E_{23}$  and  $E_{23}$  modes from step-index rectangular symmetry) showed losses 41 times higher (870 nm) and 48 times higher (1550 nm) than the fundamental. These values imply that the confinement loss would prevent higher order modes from propagating over more than a few cm, while guiding the fundamental mode over hundreds of cm. Hence, 3D PC mPOF can be considered single-moded at both 870 nm and 1550 nm. Effective single-mode behavior can also be inferred from modal field patterns (Fig. 5), which show that the second mode is not fully localized. This is evident in electric field magnitude plot and is supported by the relatively high loss value.

To investigate fiber modality further, fiber parameters were compared with [55] to check whether the fiber is in delocalized single mode or localized multimode operation region. For comparison, the hole diameter was averaged over both inner and outer ring of the actual geometry to give the following:  $d = 2.4 \mu\text{m}$ ,  $\Lambda = 5.4 \mu\text{m}$ ,  $d/\Lambda = 0.44$ . These results compared to [55] also suggest the fiber is in delocalized single mode operation at 870 nm and 1550 nm.

### C. Fiber Attenuation Measurements

The fiber attenuation was measured using the cut-back method [62]. Light from a supercontinuum source (Fianium White laser WL-SC-400) was launched via a silica fiber into one end of the 3D PC mPOF under test. The two fibers were fixed together using a ceramic ferrule. The other end of the 3D PC mPOF was placed in a second ceramic ferrule and connected directly to an optical spectrum analyzer (OSA). Two OSAs were used: a Yokogawa AQ6373B for the wavelength range 400-1200 nm and a Yokogawa AQ6370 for the 1200-2400 nm range. Transmission spectra were recorded for 7 fiber lengths, starting at 86 cm and finishing at 26 cm, giving 10 cm length decrements between measurements. In order to minimize the end face influence and maximize transmission, a few spectra were recorded for every fiber length after shortening the fiber by a small amount (1-2 mm) and the maximal transmission spectra for each length is shown in Fig. 6.

Due to the low transmission power for the longer fiber lengths only the data from the three shortest fiber lengths (26, 36, and 46 cm) was used for calculating the attenuation. The short lengths present a challenge because it then becomes difficult to remove cladding modes by bending the fiber and so

it should be noted that the data given is for all modes of the fiber. For each spectral point, the attenuation plot (Fig. 7) shows the slope resulting from the linear regression (least squares method) of the 26, 36, and 46 cm data series. In some spectral regions (1110-1205 nm, 1340-1435 nm, 1729-1747 nm), the calculated attenuation curve was very noisy due to bands of high material absorption, and hence the results in those regions are not credible. The data in the spectral region below 550 nm was discarded due to low light source intensity. The shape of the attenuation spectrum seems to align well with [13], except for wavelengths below 750 nm, which is the range that would be more susceptible to scattering losses at the air-polymer interface in the hole region.

The lowest attenuation of  $\sim 0.27$  dB/cm for 3D PC mPOF is found in a few spectral regions (780-785 nm, 820-825 nm, 953-956 nm, 1070-1090 nm). Attenuation calculated in Fig. 7 is the lowest reported in the literature so far for solid-core POFs (including step index ones) drawn from 3D-printed preform at any wavelength (see Table II). However, Table II also shows that losses for fibers drawn from mechanically casted and drilled PC preforms are up to 8 times lower, which leaves room for optimization.

Compared to simulations in Section III.-B, it can be noted that confinement loss ( $\sim 0.11$  dB/cm) accounts for  $\sim 22\%$  of total loss at 870 nm ( $\sim 0.50$  dB/cm). This ratio changes at 1550 nm, where confinement loss contributes as much as  $\sim 63\%$  ( $\sim 0.44$  dB/cm) of the total loss ( $\sim 0.69$  dB/cm) although for the measurement the noise level at 1550 nm is relatively high.

#### D. Dynamic mechanical analysis of the fiber

Dynamic mechanical analysis (DMA) was performed on the 3D PC mPOF to evaluate the Young's modulus with respect to temperature and strain cycle frequency (following standard ASTM D4065), and thermal expansion coefficient of the fiber (standard ASTM E831-14). The equipment used for the stress-strain cycles in the strain limit up to 11% was a DMA 8000 (Perkin Elmer, USA).

The first test performed was the implementation of a series of stress-strain cycles, which allowed the Young's modulus of the 3D PC mPOF at room temperature to be estimated as  $2.47 \pm 0.10$  GPa (determination coefficient  $R^2 = 0.9979$ ) – see Fig. 8. The confidence interval (CI) for measurement errors throughout this section is 95%, unless otherwise noted. The Young's modulus was estimated through the slope of the linear region of the stress-strain curve (in the range 0.05-0.3%). From Fig. 8, it is also possible to estimate yield stress, which is stress at which stress-strain curve shows non-linear behavior. Such non-linear behavior was noted for the stress higher than about 60 MPa. Yield stress is within the range for the PC material (59-70 MPa [63]), while Young's modulus slightly exceeds its upper range value (2-2.44 GPa [63]). Moreover, the Young's modulus obtained for the 3D PC mPOF is close to the one estimated for the drilled-preform 3-ring PC mPOF in [24] ( $2.89 \pm 0.30$  GPa) and [64] ( $2.70 \pm 0.01$  GPa).

In the Young's modulus characterization with respect to temperature, the strain cycle frequency used was 1 Hz, while temperature was varied from 25°C to 140°C. The strain range was 0-0.2%. The 3D PC mPOF exhibits a linear decrease of its Young's modulus with respect to a temperature increase with a slope of  $-4.775 \pm 0.094$  MPa/°C ( $R^2 = 0.9968$ ; see Fig. 9(a)).

The maximum Young's modulus variation is similar to that obtained in [24] for a drilled PC mPOF (about 0.5 GPa). However, for the 3D PC mPOF, the Young's modulus decrease shows higher linearity than in [24].

Fig. 9(b) shows the Young's modulus variation with respect to strain cycle frequency. The temperature was about 27°C, the frequency range was 0.01-10 Hz, and the strain range was 0-0.5%. The fiber exhibits an increase of Young's modulus with frequency until 5 Hz, after which sharp decrease of Young's modulus is noted. Such behavior was also observed for PMMA, Topas 5013, Zeonex 480R, and PC mPOFs in [24]. However, in the case of the 3D PC mPOF, the Young's modulus variation with respect to frequency is higher, and the slope of  $302 \pm 14$  MPa/log<sub>10</sub>(Hz) ( $R^2 = 0.9959$ ) was obtained in the range 0.01-5 Hz. This seems to contradict the results reported in [64], showing no Young's modulus dependence on frequency, which might be due to different measurement equipment.

Finally, the linear thermal expansion coefficient (linear CTE, LCTE) of the 3D PC mPOF was evaluated using TMA mode of the DMA 8000 device. A small constant tensile load ( $F = 0.01$  N) was applied on the fiber, and fiber strain was observed while changing temperature (see Fig. 10). Total variation of fiber strain ( $\epsilon_{total}$ ) with temperature was ascribed to two factors, thermal expansion ( $\epsilon_T$ ) and different response to stress caused by tensile load ( $\epsilon_\sigma$ ), coming from dependence of Young's modulus on temperature (measured earlier):

$$\epsilon_{total}(\Delta T) = \epsilon_{TE}(\Delta T) + \epsilon_\sigma(\Delta T) \quad (1)$$

In the TMA method,  $\epsilon_\sigma$  can often be neglected because it normally is much smaller than  $\epsilon_T$  due to very small value of force from tensile load. However, in the case of this fiber,  $\epsilon_{total}$  was found to be so small that neglecting  $\epsilon_\sigma$  would deviate the value of LCTE.  $\epsilon_T$  and  $\epsilon_\sigma$  can be written as:

$$\epsilon_{TE}(\Delta T) = \alpha_{//} \cdot \Delta T \quad (2)$$

$$\epsilon_\sigma(\Delta T) = \frac{\sigma}{E(\Delta T)} - \frac{\sigma}{E_0} = \frac{\sigma}{\frac{dE}{dT} \cdot \Delta T + E_0} - \frac{\sigma}{E_0} \quad (3)$$

where:  $\alpha_{//}$  (°C<sup>-1</sup>) is the linear CTE along the fiber axis,  $\Delta T$  (°C) is the temperature change above the initial temperature of the experiment ( $T_0 = 27.10 \pm 0.01$ °C),  $\sigma = F/A$  (Pa) and is the stress on the fiber coming from tensile load  $F = 0.01 \pm 0.002$  N acting on fiber area  $A$  (m<sup>2</sup>),  $E(\Delta T) = dE/dT \cdot \Delta T + E_0$  (Pa) and is the Young's modulus of the fiber,  $dE/dT$  (Pa/°C) is the dependence of Young's modulus on temperature, and  $E_0$  (Pa) is the Young's modulus at  $T_0$ . In (3), subtraction of the second term ( $\sigma/E_0$ ) is required to make  $\epsilon_\sigma$  start from zero at  $T_0$ .

By measuring the cross-sectional area of the fiber from the microscope image ( $A = 22,500 \pm 500$  μm<sup>2</sup>),  $\sigma$  was estimated to be  $4.45 \pm 0.98 \times 10^7$  Pa. This stress caused the fiber to be prestrained by  $175 \pm 39$  με at the beginning of the experiment (at  $T_0$ ), which is the value of the second term ( $\sigma/E_0$ ) in (3). Changes to  $\epsilon_\sigma$  coming from decrease of cross-sectional area of the fiber upon straining were found to be very small and hence were neglected. Based on the experiment described earlier (see Fig. 9),  $dE/dT$  and  $E_0$  were calculated to be  $-4.775 \pm 0.094$  MPa/°C and  $2.5479 \pm 0.0084$  GPa, respectively. Inserting these values to (1)-(3) allowed to plot  $\epsilon_T$  and  $\epsilon_\sigma$  along with the experimental values of  $\epsilon_{total}$  in Fig. 10. The value of LCTE along the fiber axis ( $\alpha_{//}$ ) coming from linear regression was  $7.34 \pm 0.53 \times 10^{-7}$ °C<sup>-1</sup> (CI =

95%).

This LCTE is about two orders of magnitude smaller than those usually encountered in the literature for PC material, which are in the range of  $5.0\text{-}8.1 \times 10^{-5} \text{ }^\circ\text{C}^{-1}$  (in bulk as well as in the form of thin films and pellets) [65, 66, 67, 68, 69, 70, 71]. For this reason, special care was taken to ascertain the obtained value is accurate. The calculated LCTE value was cross-validated by manual calculations to check the experimental error yielded by the total differential method and gave the result of  $7.3 \pm 1.5 \times 10^{-5} \text{ }^\circ\text{C}^{-1}$ . The larger experimental error obtained does not change the two orders of magnitude difference between the calculated value and the literature ones.

Some studies on POFs assume the literature LCTE values for bulk polymer to be valid [72, 73, 74, 75]. However, research by Zhang and Webb [76], tackling specifically the CTE of POF (made of PMMA), revealed its anisotropic behavior. According to their paper, the anisotropic nature of POF is caused by drawing process, during which polymer molecules tend to be preferentially aligned along drawing direction. This has been shown for a range of mPOFs drawn from different polymers, including PC, in [64]. Such molecular alignment in POF results in anisotropy of LCTE, making the LCTE value along drawing direction ( $\alpha_{\parallel}$ ) lower than for the isotropic material ( $\alpha_{\dots}$ ) normally reported in the literature. Consequently, the value of LCTE perpendicular to drawing direction ( $\alpha_{\perp}$ ) is higher than  $\alpha_{\dots}$ . Such behavior is well-known in the field of oriented polymers [76, 77, 78, 79], and should apply equally to POFs. In extreme cases, even negative  $\alpha_{\parallel}$  can be expected. In addition, the experimental data acquired by Zhang and Webb [76] suggested  $\alpha_{\parallel}$  to be lower than the literature values of  $\alpha_{\dots}$ . This could explain the two orders of magnitude difference in  $\alpha_{\parallel}$  obtained for the 3D PC mPOF.

#### IV. FIBER BRAGG GRATING INSCRIPTION AND TESTING

##### A. Inscription Using a Continuous-Wave Helium-Cadmium Laser

A fiber Bragg grating (FBG1) was inscribed in the manufactured PC fiber by the phase mask technique [80] using a Kimmon 325 nm CW HeCd laser (model IK3301 R-G). The laser beam (power  $\sim 23$  mW and  $\sim 1.2$  mm diameter) was focused on the fiber through a cylindrical lens of 11 cm focal length. An Ibsen Photonics custom-made phase mask of 557.50 nm pitch was placed directly on the fiber. The inscription time was 13 min. FBG1 was inscribed 10-15 mm away from the butt-coupling fiber end in order to minimize losses of optical power due to fiber attenuation. After the inscription, FBG1 was interrogated through a 3dB coupler using a supercontinuum light source Fianium White laser WL-SC-400 and a Yokogawa AQ6373B optical spectrum analyzer. The reflection spectrum recorded by OSA is shown in Fig. 11. The calculated effective refractive index of the fiber core at the Bragg wavelength  $\sim 871.8$  nm is 1.5638.

The response of FBG1 to temperature was tested. In order to facilitate grating interrogation, a silica pigtail was glued to the mPOF with FBG1. Some UV-curable glue (Norland Optical Adhesive 78) was placed in the junction point between silica and polymer fibers. They were positioned with respect to each other so that reflection spectrum was single mode, and then the

UV glue was cured. However, after curing the glue, the reflection spectrum of FBG1 displayed two peaks rather than one due to change in light coupling conditions. As it was explained in Section III.-B,  $\sim 1\text{-}1.5$  cm distance between gluing point and the POFBG allowed higher order modes to propagate. Longer distance between fiber end and the FBG would increase the loss for higher order modes greatly and render them invisible.

FBG1 was subsequently placed in a small v-groove on a Peltier plate and covered with some silicone grease to increase thermal conduction. The temperature was increased using a temperature electronic controller (TEC) from  $27^\circ\text{C}$  to  $42^\circ\text{C}$  in steps of  $5^\circ\text{C}$  in room environment. The recorded spectra are shown in Fig. 12, along with their moving averages serving to increase legibility of the plot. The temperature sensitivity was calculated from the moving averages by taking the central wavelength of -3dB bandwidth of either peak at different temperatures (see inset in Fig. 12). The calculated temperature sensitivity was around  $-21.3 \pm 1.9 \text{ pm}/^\circ\text{C}$  ( $-21.2 \pm 1.9 \text{ pm}/^\circ\text{C}$  for the left peak and  $-21.4 \pm 1.0 \text{ pm}/^\circ\text{C}$  for the right one). This value was similar to the ones achieved in previous studies on PC mPOFs:  $-25.8 \text{ pm}/^\circ\text{C}$  in a humidity-controlled environment [14] and  $-29.9 \text{ pm}/^\circ\text{C}$  in a room environment [13]. Lack of humidity control does not seem to have major impact on achieved temperature sensitivity values, as the humidity sensitivity of FBG inscribed in PC mPOF is only  $7.25 \text{ pm}/\% \text{RH}$  (as opposed to  $27.4\text{-}35.5 \text{ pm}/\% \text{RH}$  for POFs made of PMMA [26, 81]).

Knowing the temperature sensitivity of FBG1 ( $\Delta\lambda_B/\Delta T$ ), the effective RI ( $n_{eff}$ ) of the fiber core at  $\sim 871.8$  nm ( $\lambda_B$ ) and the LCTE of the fiber along fiber axis ( $\alpha_{\parallel}$ ) allowed the thermo-optic coefficient perpendicular to fiber axis ( $\text{TOC}_{\perp}$ ,  $(dn/dT)_{\perp}$ ) to be calculated. The fiber is not birefringent, so the TOC for all directions perpendicular to fiber axis is equal. COMSOL simulations showed that only up to  $\sim 3.3\%$  of electric field is parallel to fiber axis, the remainder being perpendicular to it. Hence, it can be safely assumed that the Bragg peak position depends on  $\text{TOC}_{\perp}$  only. According to the equation describing Bragg wavelength change on temperature [75]:

$$\frac{\Delta\lambda_B}{\lambda_B} = \left( \frac{1}{n_{eff}} \frac{dn}{dT} \Big|_{\perp} + \alpha_{\parallel} \right) \Delta T \quad (4)$$

thermo-optic coefficient can be expressed as:

$$\frac{dn}{dT} \Big|_{\perp} = n_{eff} \left( \frac{1}{\lambda_B} \frac{\Delta\lambda_B}{\Delta T} - \alpha_{\parallel} \right) \quad (5)$$

Using the above equation,  $\text{TOC}_{\perp}$  was calculated to be  $-39.4 \pm 3.7 \times 10^{-5} \text{ }^\circ\text{C}^{-1}$ . The experimental error was estimated with the total differential method. It should be noted that the obtained  $\text{TOC}_{\perp}$  value is less than those found in the literature for PC in bulk and in the thin film form ( $-90$  to  $-172 \times 10^{-5} \text{ }^\circ\text{C}^{-1}$ , most sources giving the value of about  $-107 \times 10^{-5} \text{ }^\circ\text{C}^{-1}$ ) [65, 66, 67, 68, 69, 82, 83, 84, 85]. Similarly to LCTE, this lower value of  $\text{TOC}_{\perp}$  can suggest that TOC is anisotropic, which would be justified in view of high anisotropy of molecular arrangement. However, further studies are required to fully understand fiber behavior.

##### B. Inscription Using a Femtosecond Laser

Another FBG (FBG2) was inscribed in the manufactured fiber by means of a femtosecond laser system (High Q Laser FemtoREGEN) operating at 517 nm with 220 fs pulse duration.

The fiber sample was fixed to a glass slide to minimize unwanted strain and bending and was subsequently mounted onto a 2D air-bearing translation system (Aerotech) for accurate motion control during inscription. The laser beam was focused on the fiber core by a long working distance objective ( $\times 50$ , Mitutoyo) using another translation stage. The laser pulse energy and repetition rate were set to  $\sim 50$  nJ/pulse and 1 kHz, respectively.

Using the plane-by-plane inscription method [86, 87, 88], 1000 planes were inscribed in the fiber core with a period of  $\sim 1.97$   $\mu\text{m}$ , giving a total grating length of 2 mm (see Fig. 13(a)).

The fiber was cleaved with a cold blade and butt-coupled to using silica fiber and refractive index matching gel. It was subsequently illuminated through a 3dB coupler using a broadband light source (ASE730, Thorlabs, range 1530-1610 nm). The reflection spectrum of the grating was measured using a commercial FBG interrogator (IBSEN IMON 512 HS), see Fig. 13(b). The effective refractive index of fiber core at  $\sim 1562$  nm ( $4^{\text{th}}$  order FBG reflection) according to the resultant reflection spectrum was estimated to be  $\sim 1.586$ .

The operation of FBG2 was tested qualitatively. Fig. 14(a) shows the response of the grating to random vibrations of the optical table after hitting it three times. The response of the grating to three breath cycles is shown in Fig. 14(b). Here, the Bragg peak wavelength decreases with the increased temperature from each exhalation, yielding negative temperature sensitivity (in line with the results achieved in Section IV.-A).

### C. Inscription Using a Krypton-Fluoride Laser

A Coherent Bragg Star Industrial-LN krypton fluoride (KrF) excimer laser system operating at 248 nm wavelength was employed to inscribe the third FBG introduced in the paper (FBG3). The pulse duration and pulse repetition rate were 15 ns and 1 Hz, respectively. The laser beam profile was measured as a rectangular Tophat function of  $6.0 \times 1.5$  mm<sup>2</sup> size and  $2 \times 1$  mrad<sup>2</sup> divergence. It was focused onto the fiber core utilizing a plano-convex cylindrical lens (Newport CSX200AR.10) with the focal length of 20 cm. The effective spot size of the beam on the fiber surface was 20.0 mm in width and 32.4  $\mu\text{m}$  in height. A slit perpendicular to the fiber direction is used to reduce the width of the beam, which defines the physical length of the grating structure (in this case 5 mm long). A 10 mm long phase mask customized for 248 nm inscription wavelength with a pitch of 567.8 nm was used for Bragg grating inscription at the 850 nm spectral region. A supercontinuum source (Fianium White laser WL-SC-400) and an optical spectrum analyzer (Yokogawa AQ6373B) were used to measure power reflected from the grating through a 3dB coupler.

Due to high fiber attenuation at this wavelength short fiber sections of around 4 cm were used. They were butt-coupled to an APC silica pigtail and a small amount of index gel was inserted to reduce Fresnel reflections. It took 5 min to inscribe a grating (300 pulses at 1 Hz pulse repetition rate, 0.6 mJ energy per pulse). The resulting FBG was 5 mm long and its Bragg wavelength was 887.25 nm (see Fig. 15). From the Bragg peak the effective refractive index at that wavelength was calculated to be 1.5626, which is very close to this calculated in Section IV.-A (1.5638 at  $\sim 871.8$  nm) and in [13].

## V. CONCLUSIONS

This paper reports the first single-mode microstructured solid-core fiber drawn from a 3D-printed preform. The fiber preform was printed in a standard lab with a commercial, low-cost printer. Whilst the preform fabrication process is not optimal in terms of the optical transparency of the fiber produced at this stage, it guides light over a sufficient distance to show the successful inscription of fiber Bragg gratings using three different laser systems: a 325 nm CW laser, a 517 nm pulsed femtosecond laser and a 248 nm pulsed nanosecond laser. The reflection peaks of these gratings were in the 850-900 nm and 1550-1600 nm regions, showing light guiding over a wide wavelength range.

Mechanical testing of the fiber showed that the 3D printing approach did not introduce any unexpected or undesirable characteristics. The fiber attenuation was the lowest reported so far for a POF drawn from a 3D-printed preform and the fiber was shown to operate in the 1550 nm region, which is useful for using with existing test and measurement equipment. The measured linear coefficient of thermal expansion was found to be two orders of magnitude smaller than for isotropic bulk material. Thermo-optic coefficient of the fiber was calculated to be only  $\sim 37\%$  of this for isotropic bulk PC. Both of these are ascribed to high anisotropy of molecular arrangement in the 3D PC mPOF, but further studies are necessary to fully understand fiber behavior.

Whilst the current performance is not yet as good as that of fibers drawn from preforms made by drilling or capillary stacking, the ultimate aim of this work is to optimize a technique that will facilitate complex fiber designs that cannot be easily fabricated with the conventional techniques. It is reasonable to expect that printing with tailor-made filaments rather than commercially available ones would allow much better control of the printing process. The custom-made filaments could be fabricated of optical quality plastics, which could result in decreasing the loss levels. In addition, their chemical composition could be precisely controlled, which could allow avoiding undesired chemical species that can adversely interfere with the POF fabrication process.

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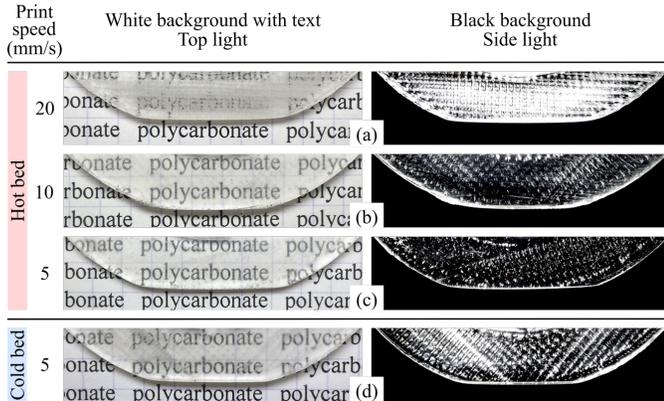


Fig. 1. Comparison of transparency of polished circular test printouts for different print speeds and printing bed temperatures (hot bed (105°C), cold bed (room temperature)). (Left) Photographs of the samples with front lighting and resting on white squared paper (5 mm pitch) with text illustrate the variable transparency of the samples. (Right) Photographs of the same samples on a black background illuminated from the side (the bottom edge of the pictures). This lighting accentuates the density and localization of air voids in the samples.

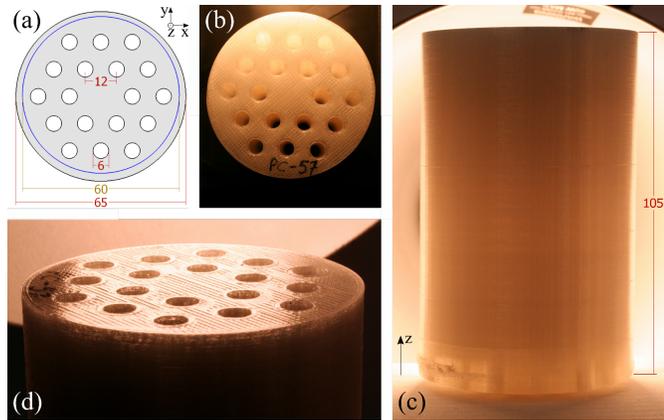


Fig. 2. (a) Top view of the preform design. The blue ring shows the machined diameter. (b) Top view of the printed preform (illuminated from the top edge). (c) Side view of the printed preform (illuminated from behind). (d) View of the top surface of the printed preform. All dimensions in mm.

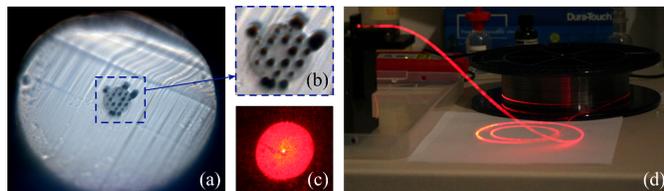


Fig. 3. (a) View of a cleaved fiber facet (blade temperature = 75°C; fiber outer diameter = 170  $\mu\text{m}$ , fiber core diameter = 8  $\mu\text{m}$ ). (b) Magnified view of the fiber core. (c) Image of a fiber end projected on screen (fiber length of 38.1 mm) with clear core guidance. See Fig. 4 for the experimental setup. (d) Photograph of butt-coupling to the fiber on spool.

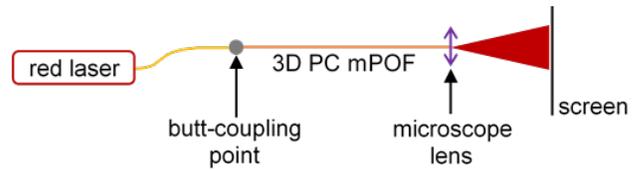


Fig. 4. Schematic showing the experimental setup used for checking core guidance (fiber length = 38.1 mm). The image observed on the screen is shown in Fig. 3(c).

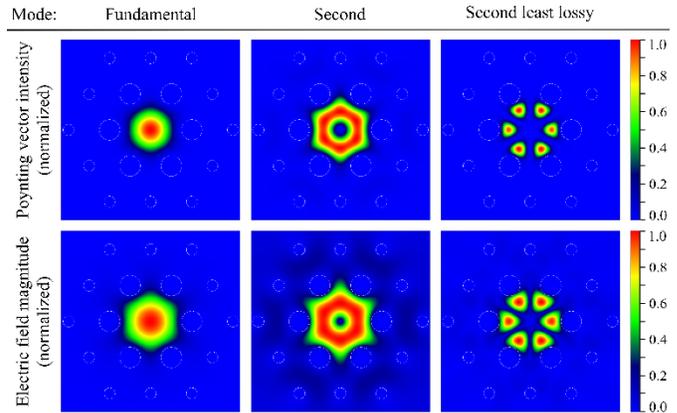


Fig. 5. Poynting vector intensity and electric field magnitude for the fundamental, second and second least lossy modes simulated for 3D PC mPOF at 870 nm in CUDOS.

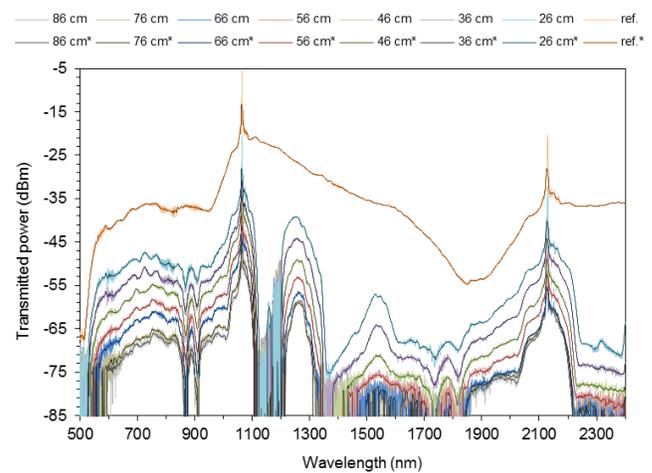


Fig. 6. Transmission spectra for different lengths of the same piece of 3D PC mPOF used for cut-back attenuation calculations. Reference spectrum (marked "ref.") comes from the supercontinuum light source attached directly to OSA. In order to reduce noise, moving averages (darker lines, data series marked with \*) have been plotted over original spectra (lighter lines). In this paper, moving averages for each spectral point have been calculated using 5 neighbors to its left and right, resulting in 11 spectral points being averaged over.

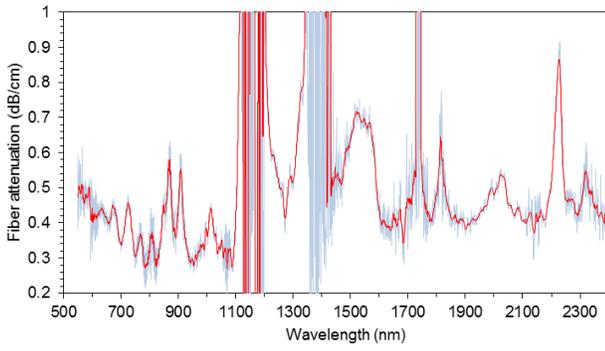


Fig. 7. Attenuation plot of the 3D PC mPOF calculated based on the transmission spectra for the three shortest fiber lengths in Fig. 6 (26, 36, and 46 cm). Moving average (red) of the attenuation has been plotted over the original spectrum (blue) to reduce noise.

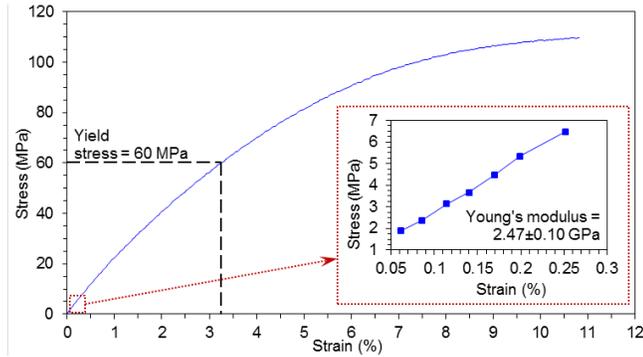


Fig. 8. Stress-strain cycles and Young's modulus of the 3D PC mPOF.

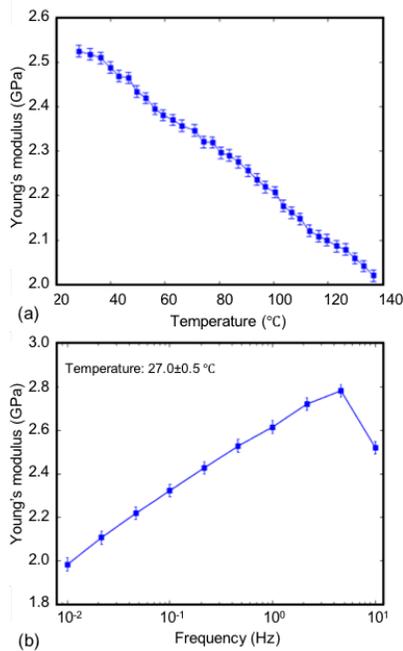


Fig. 9. Young's modulus variation for the 3D PC mPOF as a function of temperature (a) and strain cycle frequency (b). Error bars show mean deviation.

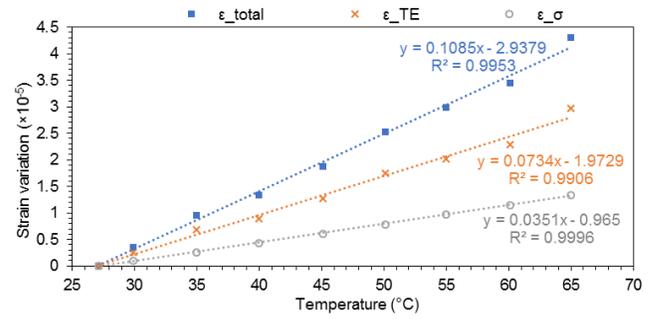


Fig. 10. Strain variation of the 3D PC mPOF ( $\epsilon_{total}$ ) as a function of temperature. It was split into strain due to thermal expansion ( $\epsilon_{TE}$ ) and different response to stress caused by tensile load of 0.01 N, coming from dependence of Young's modulus on temperature ( $\epsilon_{\sigma}$ ). See (1)-(3) for details.

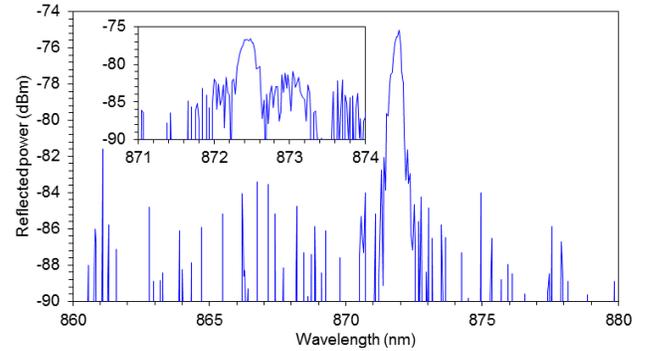


Fig. 11. Reflection spectrum of an FBG (FBG1) inscribed in the 3D PC mPOF using HeCd 325 nm laser. Depending on light coupling conditions, reflection spectrum can display one (main plot) or more peaks (inset).

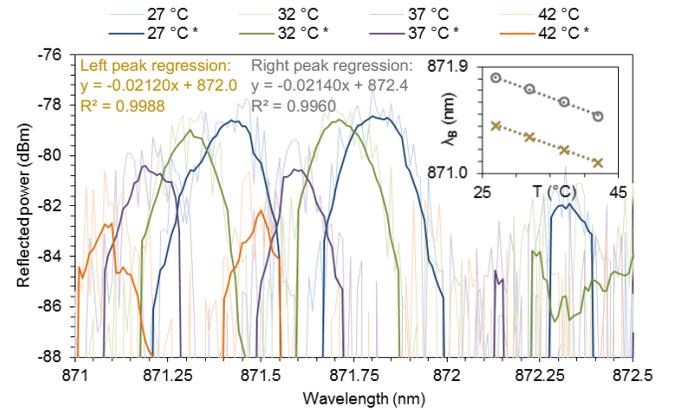


Fig. 12. Spectra of FBG1 recorded at different temperatures. Note two peaks in each spectrum corresponding to two different modes. Temperature sensitivity calculations performed on both peaks gave almost identical results (inset). In order to reduce noise, moving averages (darker lines, data series marked with \*) have been plotted over original spectra (lighter lines).

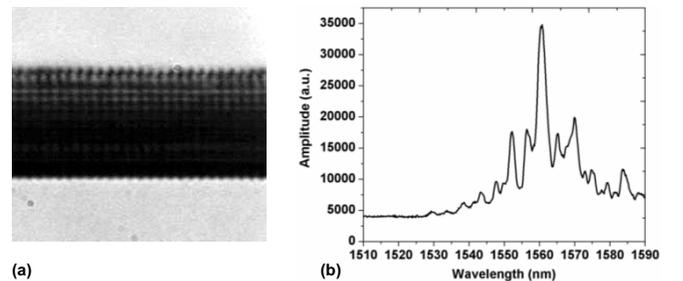


Fig. 13. (a) Microscopic image (side view) of the FBG inscribed in 3D PC mPOF by femtosecond laser (FBG2). (b) Reflection spectrum (4<sup>th</sup> order) of FBG2.

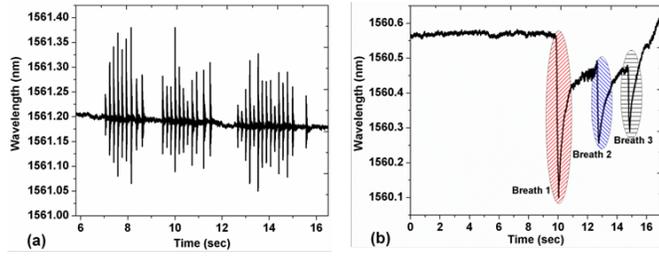


Fig. 14. (a) Time response of FBG2 to vibration induced by hitting the optical table. (b) Time response of the FBG to three breathing shots, resulting in pulses of warmer and more humid air. As shown earlier, the POFBG shows negative temperature sensitivity. The Bragg wavelength increase after  $t = 15$  s in (b) might be due to positive humidity sensitivity of the fiber [14].

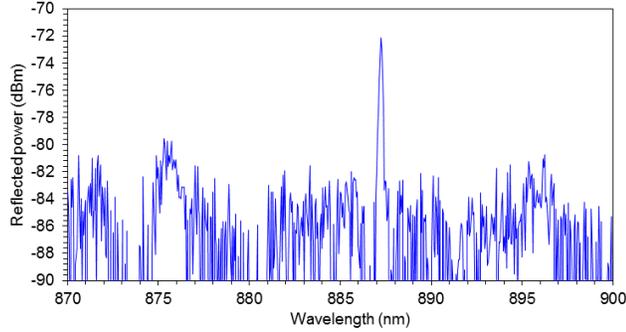


Fig. 15. Reflection spectrum of a 248nm KrF laser-inscribed FBG in the 3D PC mPOF.

Table I.  
RESULTS OF CUDOS SIMULATIONS FOR FUNDAMENTAL, SECOND AND SECOND LEAST LOSSY MODES FOR 870 NM AND 1550 NM

Mode	Fundamental	Second	Second least lossy
Mode class	3/4 (degenerate)	2 (non-degenerate)	5/6 (degenerate)

Table II.  
COMPARISON OF ATTENUATION (DB/CM) OF SOLID-CORE POFs FROM 3D-PRINTED AND DRILLED PREFORMS<sup>a</sup>

Wavelength (nm)	3D-printed preforms		Drilled preforms			
	SI-POF [4] (PC+ABS)	Air-structured POF [5] (SBP)	SI-POF [3], (SBP+PETG)	3D PC mPOF (this paper, PC) <sup>b</sup>	mPOF [13] (PC)	Optimized mPOF [14] (PC)
543			0.64	0.53		
632-633	0.7-1.8 <sup>c</sup>	~1.5		0.44	0.16	0.055
819				0.29		0.041
833.5				0.32	0.089	
1047-1052			0.44	~0.34 <sup>d</sup>		
1064		~0.75		0.33		
1520-1560			0.94	~0.68 <sup>d</sup>		
1550		~1.51		0.69		

<sup>a</sup>Fiber materials are given in brackets after fiber name; for step-index (SI) fibers, both core and cladding materials, respectively, are given. <sup>b</sup>For full attenuation spectrum, see Fig. 7. <sup>c</sup>Attenuation depending on 3D-printed core shape: square core – 1.2 dB/cm; triangular core – 1.3 dB/cm; rectangular core – 1.8 dB/cm. Attenuation of 0.7 dB/cm measured for circular core of solid material (not 3D-printed). <sup>d</sup>Average values for the wavelength range.

Step-index fiber equivalent	HE <sub>11</sub> (circular symmetry)	TE <sub>01</sub> (circular symmetry)	E <sup>x/y</sup> <sub>23</sub> (rectangular symmetry)
Wavelength = 870 nm			
Re( $n_{eff}$ )	1.57535279891	1.57296137794	1.56379542212
Im( $n_{eff}$ )	$1.75892967786 \times 10^{-7}$	$3.86335837268 \times 10^{-5}$	$7.15216328426 \times 10^{-6}$
Confinement loss (dB/cm)	0.11033757	24.23482671	4.48654826
Wavelength = 1550 nm			
Re( $n_{eff}$ )	1.56194339477	1.55462887006	1.52603245572
Im( $n_{eff}$ )	$1.23636037073 \times 10^{-6}$	$1.50868839412 \times 10^{-4}$	$5.90364581028 \times 10^{-5}$
Confinement loss (dB/cm)	0.4353189302	53.1204843936	20.7865670839

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